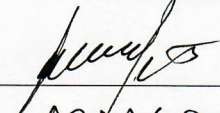
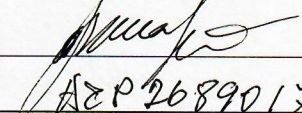


	East Coast Aircraft Painting, Inc. DeLand Municipal Airport (DED) 2007 Industrial Drive Deland, FL 32724 Phone: (855) 519-3067	
A/C MAKE: Piper Aircraft	MODEL: PA 46-350P	AC REGISTRATION #: N777WZ
SERIAL #: 4636491	DATE: 02/23/2026	
PRIMARY: 200737 Cloud Blue Metallic (Sherwin Williams)		
SECONDARY: 06920 Mystical Blue Metallic (Sherwin Williams)		
TRIM: CM570535 Matterhorn White (Sherwin Williams)		
TRIM: 01955 Black (Sherwin Williams)		
The aircraft was stripped, etched, primed and painted with the above colors. The flight controls surfaces primary and secondary were removed before painting of aircraft. After painting primary flight controls were checked for balance then both primary and secondary were reinstalled back on aircraft. Replaced 8 static wick part number 456-904 and bases part number 452-124. Removed previous registration number N263SP and installed a new registration number N777WZ on both sides of the fuselage between the trailing edge of the wing and the leading edge of the horizontal stabilizer. Registration number dimensions found in accordance with 14CFR Part 45.29. All paperwork in file. All work was performed to maintenance manual. With respect to the above work performed, this aircraft is approved for return to service. END.		

Signature 

License # AEP2689017

	East Coast Aircraft Painting, Inc. DeLand Municipal Airport (DED) 2007 Industrial Drive Deland, FL 32724 Phone: (855) 519-3067	
A/C MAKE: Piper Aircraft	MODEL: PA 46-350P	AC REGISTRATION #: N777WZ
SERIAL #: 4636491	DATE: 02/23/2026	
Recovered seats with Perrone's Duchess Spun Silver, leather and Wollsdorf's Ambassador Dark Grey, leather.		
Rewebbed seatbelts with Aviation Safety's Moondust.		
Recovered headliner with Douglass Interior's Columbia Ultraleather Glacier and Shadow.		
Recovered window panel with Douglass Interior's Columbia Ultraleather Glacier.		
Recovered lower side panel with Perrone's Duchess Spun Silver, leather and Wollsdorf's Ambassador Dark Grey, leather.		
Replaced carpet with Hafner Textile's Aurora 761 Charcoal.		
Recovered baggage compartment with Tapis's Grospoint Black.		
Recovered steps with Aircraft Interior Product's Aermat Black.		
Recovered table cabinetry with Wilsonart's Asian Night Gloss.		
All materials include burn test certification, copies provided to owner/operator. All maintenance performed on this aircraft was performed in accordance with current regulations of the FAA and was determined to be in an airworthy condition and is approved for return to service with respect to the work performed. END.		

Signature 

License # AEP2689017



FLAMMABILITY TEST RESULTS

U.S. TECHNICAL CONSULTANTS, INC.

2467 E. ORANGETHORPE AVE.

FULLERTON, CA. 92831

714-446-1920

www.ustechnical.com

VERTICAL

CUSTOMER	Douglass Interior Products	TEST DATE	12/14/2023
CUSTOMER PO	45773	TEST DATA SHEET RECORD No.	23569-1
WORK ORDER	23569	TEST PROCEDURE	N/A
FAA PROJECT No.	N/A	PROCEDURE REVISION	N/A

TEST SPECIMEN DESCRIPTION

REQUIREMENTS:

LENGTH: 12.0 inches WIDTH: 3.0 inches

THICKNESS: No thicker than the minimum thickness to be qualified for use in airplane, and foams: 0.5 inch thick

DESCRIPTION	Glacier Columbia	PART NUMBER	DEF-CD287	LOT NUMBER	1159-02
LENGTH	12	WIDTH	3	THICKNESS	.03

TEST SPECIMEN CONDITIONING

REQUIREMENT:

70°F ± 5°F (21°C ± 3°C) and at 50% ± 5% relative humidity until moisture equilibrium is reached or 24 hours

START DATE AND TIME	12/13/2023 1:00 PM	END DATE AND TIME	12/14/2023 1:00 PM
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TEST METHOD

METHOD /
IGNITION TIME

RELEVANT 14 CFR REGULATION(S) COMPLIED WITH

☐ VERTICAL, 60 SECOND 14 CFR Part 25 Appendix F Part II(a)(1)(i)	☐ Part 23 §23.853 Amendment 23-62, §23.855 Amendment 23-49, §23.2325 Amendment 23-64 ☐ Part 25 §25.853, §25.855	☐ Part 27 §27.853, §27.855 ☐ Part 29 §29.853
☒ VERTICAL, 12 SECOND 14 CFR Part 25 Appendix F Part II(a)(1)(ii), (iii), (a)(2)(iv)	☐ Part 23 §23.853 Amendment 23-62, §23.855 Amendment 23-49, §23.2325 Amendment 23-64 ☒ Part 25 §25.853, §25.855	☐ Part 27 §27.853, §27.855 ☐ Part 29 §29.853

FLAME TEMPERATURE

REQUIREMENT Minimum Temperature: 1550°F(843.33°C)

TEMPERATURE VERIFICATION



TEST RESULTS

60 SECOND TEST		Self	BURN LENGTH:		AFTER-FLAME TIME:		DRIP FLAME TIME:				
RESULT REQUIREMENTS:		Extinguishing	≤ 6.0 inches		≤ 15 seconds		≤ 3 seconds				
12 SECOND TEST		Self	BURN LENGTH:		AFTER-FLAME TIME:		DRIP FLAME TIME:				
RESULT REQUIREMENTS:		Extinguishing	≤ 8.0 inches		≤ 15 seconds		≤ 5 seconds				
			IN FABRIC: WARP AND FILL DIRECTION OF WEAVE MUST BE TESTED								
			WARP				FILL (WEFT)				
SPECIMEN NUMBER	BURN LENGTH	AFTER-FLAME TIME	DRIP FLAME TIME	SPECIMEN NUMBER	BURN LENGTH	AFTER-FLAME TIME	DRIP FLAME TIME	SPECIMEN NUMBER	BURN LENGTH	AFTER-FLAME TIME	DRIP FLAME TIME
				1	3	19	0	1	4.5	19	0
				2	4	8	0	2	3.25	2	0
				3	3.75	12	0	3	4	18	0
AVERAGE				AVERAGE	3.58	13	0	AVERAGE	3.91	13	0

TEST CONCLUSION

☒ PASS

☐ FAIL

NOTES:

Alexander Petrisor NAME (PRINT)	TESTED BY <i>Alex Petrisor</i> SIGNATURE	12/14/2023 DATE
Francisco Valerio NAME (PRINT)	WITNESSED BY VIDEO <i>[Signature]</i> SIGNATURE	12/14/2023 DATE



FLAMMABILITY TEST RESULTS

U.S. TECHNICAL CONSULTANTS, INC.

2467 E. ORANGETHORPE AVE.

FULLERTON, CA. 92831

714-446-1920

www.ustechnical.com

VERTICAL

CUSTOMER	Douglass Interior Products	TEST DATE	3/27/2025
CUSTOMER PO	46701	TEST DATA SHEET RECORD No.	25143-1
WORK ORDER	25143	TEST PROCEDURE	N/A
FAA PROJECT No.	N/A	PROCEDURE REVISION	N/A

TEST SPECIMEN DESCRIPTION

REQUIREMENTS:

LENGTH: 12.0 inches

WIDTH: 3.0 inches

THICKNESS: No thicker than the minimum thickness to be qualified for use in airplane, and foams: 0.5 inch thick

DESCRIPTION	Shadow Columbia	PART NUMBER	DEF-CD-BR831	LOT NUMBER	1175-01
LENGTH	12	WIDTH	3	THICKNESS	.03

TEST SPECIMEN CONDITIONING

REQUIREMENT:

70°F ± 5°F (21°C ± 3°C) and at 50% ± 5% relative humidity until moisture equilibrium is reached or 24 hours

START DATE AND TIME	3/26/2025 3:00 PM	END DATE AND TIME	3/27/2025 3:00 PM
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TEST METHOD

METHOD /
IGNITION TIME

RELEVANT 14 CFR REGULATION(S) COMPLIED WITH

☐ VERTICAL, 60 SECOND 14 CFR Part 25 Appendix F, Part I(a)(1)(i)	☐ Part 23 §23.853 Amendment 23-62, §23.855 Amendment 23-49, §23.2325 Amendment 23-64 ☐ Part 25 §25.853, §25.855	☐ Part 27 §27.853, §27.855 ☐ Part 29 §29.853
☒ VERTICAL, 12 SECOND 14 CFR Part 25 Appendix F, Part I(a)(1)(ii), (iii), (a)(2)(iv)	☐ Part 23 §23.853 Amendment 23-62, §23.855 Amendment 23-49, §23.2325 Amendment 23-64 ☒ Part 25 §25.853, §25.855	☐ Part 27 §27.853, §27.855 ☐ Part 29 §29.853

FLAME TEMPERATURE

REQUIREMENT Minimum Temperature: 1550°F(843.33°C)

TEMPERATURE VERIFICATION ☒

TEST RESULTS

60 SECOND TEST		Self	BURN LENGTH:		AFTER-FLAME TIME:		DRIP FLAME TIME:				
RESULT REQUIREMENTS:		Extinguishing	≤ 6.0 inches		≤ 15 seconds		≤ 3 seconds				
12 SECOND TEST		Self	BURN LENGTH:		AFTER-FLAME TIME:		DRIP FLAME TIME:				
RESULT REQUIREMENTS:		Extinguishing	≤ 8.0 inches		≤ 15 seconds		≤ 5 seconds				
			IN FABRIC: WARP AND FILL DIRECTION OF WEAVE MUST BE TESTED								
			WARP				FILL (WEFT)				
SPECIMEN NUMBER	BURN LENGTH	AFTER-FLAME TIME	DRIP FLAME TIME	SPECIMEN NUMBER	BURN LENGTH	AFTER-FLAME TIME	DRIP FLAME TIME	SPECIMEN NUMBER	BURN LENGTH	AFTER-FLAME TIME	DRIP FLAME TIME
				1	2	0	0	1	1	0	0
				2	1.25	0	0	2	2	0	0
				3	2	0	0	3	1.25	0	0
AVERAGE				AVERAGE	1.75	0	0	AVERAGE	1.41	0	0

TEST CONCLUSION ☒ PASS ☐ FAIL

NOTES:

TESTED BY		
Alexander Petrisor		3/27/2025
NAME (PRINT)	SIGNATURE	DATE
WITNESSED BY VIDEO		
Jonathan Prieto		3/27/2025
NAME (PRINT)	SIGNATURE	DATE



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VERTICAL FLAMMABILITY TEST RESULTS		Skandia, Inc WO# 432337		
14 CFR Part 25.853 (a) Amdt 25-116 Appendix F Part I (a)(1)(ii)		Client PO 090225		
Perrone Performance Leather & Textiles 182A Riverside Drive Fultonville NY 12072		Doc ID 12NW-529856 Test Plan # Rev Project # Technician Armando Anaya		
AIRCRAFT OR AIRCRAFT COMPONENT IDENTIFICATION				
Make: Stock		Model: Serial: Tail:		
Conditioning Room Data: Date In: 09/04/2025 Time In: 11:05 AM Date Out: 09/05/2025 Time Out: 1:05 PM				
SPECIMEN MATERIALS				
Part No: DUC-2657 Perrone Performance Leathers & Textiles: Duchess, Spun Silver				
Lot No: 855157254				
Test Set	Flame Application (seconds)	Flame Time (seconds)	Burn Length (inches)	Drippings (seconds)
1	12	22.1	2.3	0.0
2	12	9.7	1.9	0.0
3	12	3.0	1.4	0.0
Average:		11.6	1.9	0.0
Vertical (12 second) Burn Test: Average Self-Extinguish time may not exceed 15 seconds. Average Burn Length may not exceed 8 inches. Average Dripping may not exceed 5 seconds after falling.				
Comments:				
☒ Passed ☐ Failed				
CERTIFICATION: I certify that after testing these specimens, the above results were obtained in accordance with the procedures and equipment specified by Code of Federal Regulations Title 14 Part 25, or the Aircraft Materials Fire test handbook. Not FAA approved unless an 8110-3 has been issued.				
Armando Anaya		Digitally signed by Armando Anaya Date: 2025.09.05 13:09:27 -05'00'		09/05/2025
WITNESS		DATE		

Test results relate only to the items tested. Results apply to the sample as received (as applicable).
Uncertainty is not taken into account in the statement of conformity in relation to the test results.



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VERTICAL FLAMMABILITY TEST RESULTS		Skandia, Inc WO# 411306		
14 CFR Part 25.853 (a) Amdt 25-116 Appendix F Part I (a)(1)(ii)		Client PO		
Wollsdorf Leder Wollsdorf 80 Wollsdorf 8181 AT		Doc ID 12NW-502932 Test Plan # Rev Project # Technician Zach Boehning		
AIRCRAFT OR AIRCRAFT COMPONENT IDENTIFICATION				
Make: Stock		Model: Serial: Tail:		
Conditioning Room Data: Date In: 11/20/2023 Time In: 1:05 PM Date Out: 11/21/2023 Time Out: 1:14 PM				
SPECIMEN MATERIALS				
Part No: 04500062/15065 Wollsdorf Leder Austria: Ambassador, Dark Grey				
Lot No: 485246				
Test Set	Flame Application (seconds)	Flame Time (seconds)	Burn Length (inches)	Drippings (seconds)
1	12	0.0	0.3	0.0
2	12	0.0	0.3	0.0
3	12	0.0	0.2	0.0
Average:		0.0	0.3	0.0
Vertical (12 second) Burn Test: Average Self-Extinguish time may not exceed 15 seconds. Average Burn Length may not exceed 8 inches. Average Dripping may not exceed 5 seconds after falling.				
Comments:				
☒ Passed ☐ Failed				
CERTIFICATION: I certify that after testing these specimens, the above results were obtained in accordance with the procedures and equipment specified by Code of Federal Regulations Title 14 Part 25, or the Aircraft Materials Fire test handbook. Not FAA approved unless an 8110-3 has been issued.				
Zachary Boehning		Digitally signed by Zachary Boehning Date: 2023.11.21 13:16:17 -06'00'		11/21/2023
WITNESS		DATE		

Test results relate only to the items tested. Results apply to the sample as received (as applicable).
Uncertainty is not taken into account in the statement of conformity in relation to the test results.

1. Approved Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. FORM TRACKING NUMBER: 121125 - 13	
4. Organization Name and Address: AVIATION SAFETY PRODUCTS, INC. 41 EASTERLING RD BLAIRSVILLE, GA 30512				5. WorkOrder/Contract/Invoice Number: 32130-1 (XWJR831X)	
6. Item: 01	7. Description: Restraint	8. Part Number: 5-01-475768	9. Quantity: 1	10. Serial Number: NA	11. Status/Work: Repaired
12. Remarks: Rewebbed and Inspected Full assembly; Piper P/N 564-912 The work described above was performed in accordance with Federal Aviation Administration regulations and Aviation Safety Products Inc. proprietary FAA Approved Repair Specification ASRPS1 Rev 1 dated 09/03/2012 The installer is responsible for any applicable Airworthiness Directives.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Brandy Henson</i>	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: XWJR831X	
				14d. Name (typed or Printed): Brandy Henson	
				14e. Date (dd/mm/yyyy): 24/Dec/2025	

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/Installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts engin(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approved Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. FORM TRACKING NUMBER: 121125 - 14	
4. Organization Name and Address: AVIATION SAFETY PRODUCTS, INC. 41 EASTERLING RD BLAIRSVILLE, GA 30512 (XWJR831X)				5. WorkOrder/Contract/Invoice Number: 32130-2	
6. Item: 01	7. Description: Restraint	8. Part Number: S/5-01-1V5768	9. Quantity: 1	10. Serial Number: NA	11. Status/Work: Repaired
12. Remarks: Rewebbed and Inspected Tip and inertia reel portion for Type 5-01-1V5768. Piper P/N 564-910 The work described above was performed in accordance with Federal Aviation Administration regulations and Aviation Safety Products Inc. proprietary FAA Approved Repair Specification ASRPS1 Rev 1 dated 09/03/2012 The installer is responsible for any applicable Airworthiness Directives.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Brandy Henson</i>	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: XWJR831X	
				14d. Name (typed or Printed): Brandy Henson	
				14e. Date (dd/mm/yyyy): 24/Dec/2025	

User/Installer Responsibilities

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Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approved Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. FORM TRACKING NUMBER: 121125 - 15	
4. Organization Name and Address: AVIATION SAFETY PRODUCTS, INC. 41 EASTERLING RD BLAIRSVILLE, GA 30512				5. WorkOrder/Contract/Invoice Number: 32130-3 (XWJR831X)	
6. Item: 01	7. Description: Restraint	8. Part Number: S/5-01-1P5768	9. Quantity: 1	10. Serial Number: NA	11. Status/Work: Repaired
12. Remarks: Rewebbed and Inspected Tip and inertia reel portion for Type 5-01-1P5768. Piper P/N 564-908 The work described above was performed in accordance with Federal Aviation Administration regulations and Aviation Safety Products Inc. proprietary FAA Approved Repair Specification ASRPS1 Rev 1 dated 09/03/2012 The installer is responsible for any applicable Airworthiness Directives.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Brandy Henson</i>	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: XWJR831X	
				14d. Name (typed or Printed): Brandy Henson	
				14e. Date (dd/mm/yyyy): 24/Dec/2025	

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/Installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approved Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. FORM TRACKING NUMBER: 121125 - 16	
4. Organization Name and Address: AVIATION SAFETY PRODUCTS, INC. 41 EASTERLING RD BLAIRSVILLE, GA 30512 (XWJR831X)				5. WorkOrder/Contract/Invoice Number: 32130-4	
6. Item: 01	7. Description: Restraint	8. Part Number: S/5-01-1V0768	9.Quantity: 1	10.Serial Number: NA	11. Status/Work: Repaired
12. Remarks: Rewebbed and Inspected Tip and inertia reel portion for Type 5-01-1V0768. Piper P/N 564-909 The work described above was performed in accordance with Federal Aviation Administration regulations and Aviation Safety Products Inc. proprietary FAA Approved Repair Specification ASRPS1 Rev 1 dated 09/03/2012 The installer is responsible for any applicable Airworthiness Directives.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Brandy Henson</i>	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: XWJR831X	
				14d. Name (typed or Printed): Brandy Henson	
				14e. Date (dd/mm/yyyy): 24/Dec/2025	

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/Installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approved Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. FORM TRACKING NUMBER: 121125 - 17	
4. Organization Name and Address: AVIATION SAFETY PRODUCTS, INC. 41 EASTERLING RD BLAIRSVILLE, GA 30512				5. WorkOrder/Contract/Invoice Number: 32130-5 (XWJR831X)	
6. Item: 01	7. Description: Restraint	8. Part Number: S/5-01-1P0768	9. Quantity: 1	10. Serial Number: NA	11. Status/Work: Repaired
12. Remarks: Rewebbed and Inspected Tip and inertia reel portion for Type 5-01-1P0768. Piper P/N 564-907 The work described above was performed in accordance with Federal Aviation Administration regulations and Aviation Safety Products Inc. proprietary FAA Approved Repair Specification ASRPS1 Rev 1 dated 09/03/2012 The installer is responsible for any applicable Airworthiness Directives.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Brandy Henson</i>	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: XWJR831X	
				14d. Name (typed or Printed): Brandy Henson	
				14e. Date (dd/mm/yyyy): 24/Dec/2025	

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/Installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approved Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. FORM TRACKING NUMBER: 121125 - 18	
4. Organization Name and Address: AVIATION SAFETY PRODUCTS, INC. 41 EASTERLING RD BLAIRSVILLE, GA 30512				5. WorkOrder/Contract/Invoice Number: 32130-6 (XWJR831X)	
6. Item: 01	7. Description: Restraint	8. Part Number: 5-01-470768	9. Quantity: 1	10. Serial Number: NA	11. Status/Work: Repaired
12. Remarks: Rewebbed and Inspected Full Assembly: Piper P/N 564-911 The work described above was performed in accordance with Federal Aviation Administration regulations and Aviation Safety Products Inc. proprietary FAA Approved Repair Specification ASRPS1 Rev 1 dated 09/03/2012 The installer is responsible for any applicable Airworthiness Directives.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Brandy Henson</i>	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: XWJR831X	
				14d. Name (typed or Printed): Brandy Henson	
				14e. Date (dd/mm/yyyy): 24/Dec/2025	

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/Installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

CERTIFICATE OF FLAMMABILITY TESTING

CC AVIATION, LLC.

BRIAN FOSS
2444 JOSSMAN RD.
HOLLY, MI 48442
(248) 716-9585
FAX: (888) 990-1937
E-MAIL Brian@ccaviation.net

TEST REPORT NO: 123888
CONDITIONING DATE: 08/24/25
TEST DATE: 08/25/25

TEMP: 66° HUMID: 50% FLAME 1574°
CONDITIONING TIME: 01:00 PM
TEST TIME: 01:10 PM

MATERIAL IDENTIFICATION: Carpet: Style: Aurora Color: Charcoal Dye Lot: 761-825

APPLICATION: DATA SHEET

CUSTOMER: HAFNER SPECIALTIES

12 Second Vertical Ignition: 14 CFR 25.853 (a) Amdt. 25-116 Appendix F, Part I (a) (1) (ii).
Requirements: 15 sec. extinguish time: 8 inch burn length: 5 sec drip extinguish time:

	Extinguish Time: 0.0	Burn Length: 0.1	Drip Extinguish Time: None
	Extinguish Time: 0.0	Burn Length: 0.1	Drip Extinguish Time: None
	Extinguish Time: 0.0	Burn Length: 0.1	Drip Extinguish Time: None
	Extinguish Time:	Burn Length:	Drip Extinguish Time:
	Extinguish Time:	Burn Length:	Drip Extinguish Time:
	Extinguish Time:	Burn Length:	Drip Extinguish Time:
	AVERAGE 0.00	AVERAGE 0.10	AVERAGE 0.00

COMMENTS:

PASSED: X

FAILED:

Testing Performed and Written By: Derek Betts

Witnessed and Approved By: Brian Foss Engineering Designee: Structures (Flammability Only)



VERTICAL FLAMMABILITY TEST OBSERVATION SHEET

Test Date: February 25, 2025

Flame Calibration Temp (min 1550°F): 1568 °F

Customer: AIRCRAFT INTERIOR PRODUCTS

Purchase Order: ST436860001

Report #: AIP022025-0002

Description/Material: AERMAT (45R/1000/30R/54, BLACK) – VINYL / LOT: J100029438.0000

Conditioning Chamber Date: February 20, 2025

Time: 4:50 PM

Conditioning Requirements: 24 Hours @ 70 ± 5°F & 50% ± 5% RH

Conditioning Checked By: BH

Test Type:	☒ 12-Second Vertical	☐ 60-Second Vertical
Federal Aviation Regulation(s)	14 CFR § 23.853(d)(3)(ii) Amdt. 23-62 14 CFR § 25.853(a) Amdt. 25-116 14 CFR § 29.853(a)(2) Amdt. 29-23	14 CFR § 23.853(d)(3)(i) Amdt. 23-62 14 CFR § 25.853(a) Amdt. 25-116 14 CFR § 29.853(a)(1) Amdt. 29-23
Pass/Fail Requirements	After Flame Time Avg: May not exceed 15.0 seconds. Burn Length Avg: May not exceed 8.0". Drip Flame Time Avg: May not exceed 5.0 seconds.	After Flame Time Avg: May not exceed 15.0 seconds. Burn Length Avg: May not exceed 6.0". Drip Flame Time Avg: May not exceed 3.0 seconds.

NOTES/SPECIAL INSTRUCTIONS

TEST RESULTS

MATERIAL DIRECTION	SPECIMEN #	AFTER FLAME TIME (Sec)	BURN LENGTH (in)	DRIP FLAME TIME (Sec)	COMMENTS
☐ Warp ☐ Fill ☒ N/A	1	0.00	1.30	0.00	Drips: None
	2	0.00	1.20	0.00	Drips: None
	3	0.00	1.60	0.00	Drips: None
	AVERAGE	0.00	1.37	0.00	
☐ Warp ☐ Fill ☐ N/A	1	-	-	-	Drips:
	2	-	-	-	Drips:
	3	-	-	-	Drips:
	AVERAGE	0.00	0.00	0.00	
PASS	FAIL	TIME		CONDUCTED BY	COGNIZANT DER
X		9:58 AM		Bryan Hartsel	Paul Nething

The above test results were obtained in accordance with the procedures and equipment specified by the test type identified above or the Aircraft Materials Fire Test Handbook. Not FAA approved unless an 8110-3 has been issued.

FLAMMABILITY TEST DATA SHEET

DATE: 4/3/2025 PN#: 9001613 PROJECT REF: PATTERN: GROSPPOINT PN#: GPFRISG0094 COLOR: BLACK DYE LOT: Z ROLL#: 765, 764-799				Test(s) Requested: ☒ 12 sec. Vertical Burn FAR 25.853 (a) Appendix F, Part I, (a), (ii) ☐ 60 sec. Vertical Burn FAR 25.853 (a) Appendix F, Part I, (a), (i) ☐ 60 sec. Vertical Burn 14 CFR 25, Appendix F, Part I ☐ 15 sec. Horizontal Burn FAR 25.853 (a) Appendix F, Part I, (a), (iv) ☐ OSU Heat Release FAR 25.853 (d) Appendix F, Part IV ☐ Smoke Density FAR 25.853 (d) Appendix F, Part V / ASTM E662-01			
Tested by: Erika Martinez 		Witnessed by: N/A		Conditioning (24hr. @ 70°F ±5°F, 50% ±5% RH) In: 4/1/2025 4:50PM Out: 4/3/2025 10:13AM			

12 Second Vertical Burn Test				60 Second Vertical Burn Test			
SAMPLE NO.	EXTINGUISHING TIME (sec.)	BURN LENGTH (in.)	DRIP EXT. TIME (sec.)	SAMPLE NO.	EXTINGUISHING TIME (sec.)	BURN LENGTH (in.)	DRIP EXT. TIME (sec.)
1	0.0	1.1	0.0	1			
2	0.0	1.2	0.0	2			
3	1.3	1.2	0.0	3			
4				4			
AVG.	0.4	1.2	0.0	AVG.			
P/F	Pass ☒ Fail ☐			P/F	Pass ☐ Fail ☐		

15 Second Horizontal Burn Test				COMMENTS: Quality Assurance Only LIMITS: Self Extinguishing - 15 sec. Max. Burn Length - 8.0 in. (12 Sec.)/6.0 in. (60 Sec.) Drip Extinguishing - 5.0 sec. (12 Sec.)/3.0 sec.(60 Sec.) Burn Rate = Burn length (in.) / Burn Time (min.)
SAMPLE NO.	BURN RATE (in./min.)	BURN LENGTH (in.)	BURN TIME (sec.)	
1				
2				
3				
AVG.				
P/F	Pass ☐ Fail ☐			